

## **ASTP RBCC Activities**

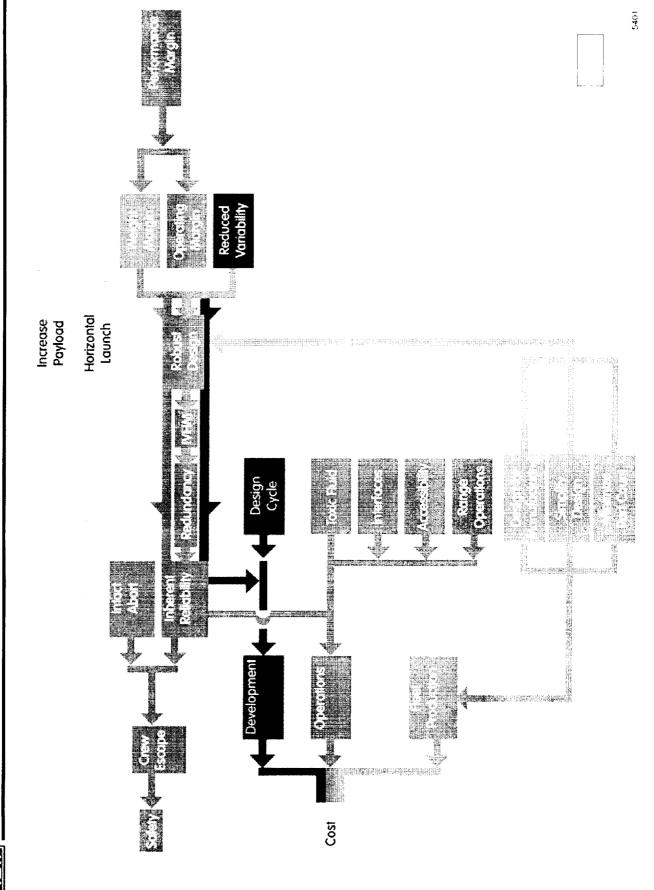
Karl W. Nelson Marshall Space Flight Center 10/11/00

#### **Process** 3rd Generation Technology Prioritization



# ◆ Inter-center 3rd Generation team ranked over 70 technologies

- Potential payoff to cost and safety
- Technological risk
- Ranking process included study of technology activity, open discussion and consensus
- Analytical Hierarchy Process (AHP) used in final ranking
- Technologies prioritized by
- A combined score of potential system payoff and technological risk Technology score = (cost + safety)\* risk°
- Technologies coordinated with Aero-Space base activities
- Efforts underway to make this annual process



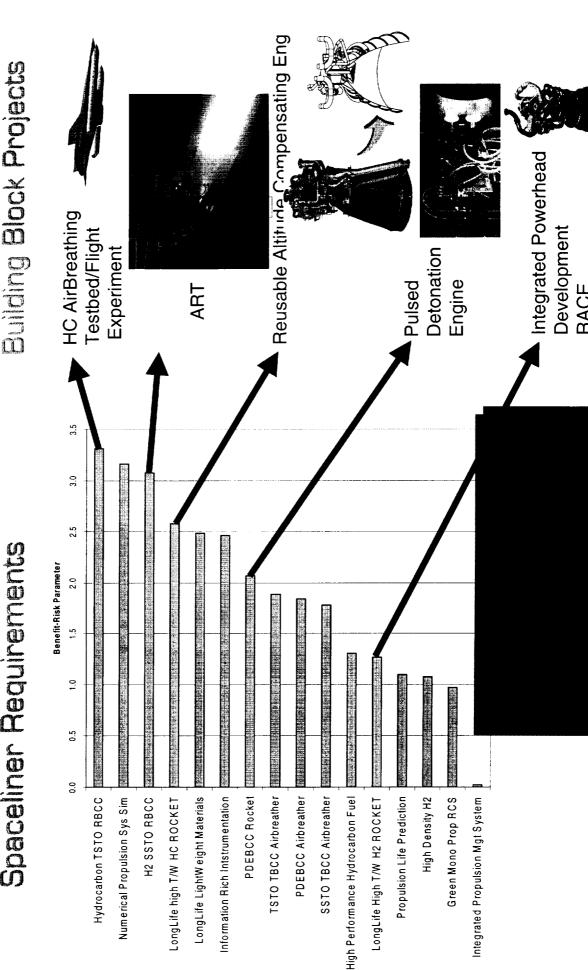


## Example 3rd Gen. Propulsion Candidate

Investments



### Spaceliner Requirements



## Hydrogen RBCC



### Recent Accomplishments



## ART (Advanced Reusable Technologies)

- Aerojet & Rocketdyne Flowpath Tested
- Test Conducted From M 0 to Mach 8
- Total Of 253 Test Conducted
- Good Overall Performance

#### Several Firsts In Testing

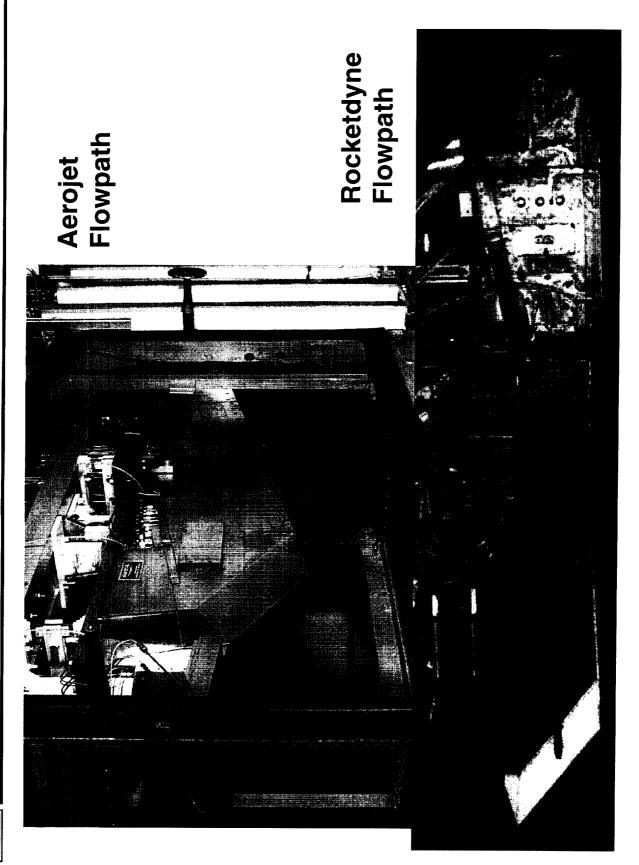
- Dynamic Trajectory Simulation (AAR -> RAM and RAM-> SCRAM))
- · SCRAM Testing @ High Dynamic Pressure (M8 @ 1,200 Psf)
- · Rocketdyne A-5 engine has logged over 1 hour of accumulated test time
- Parametric Test Performed By Pennsylvania State University

### Trailblazer Concept Development

- Lead By Glenn Research Center
- Currently Testing @ GASL

#### **System Studies**

- Various Vehicle/Engine Combinations Being Studied
- RBCC
- TBCC
- PDE
- Sensitivity Trades Being Made
- Trajectories
- Fineness ratio
- Payload capability









Sea-Level Static	31	173	34	342
Air-Augmented Rocket	12	26	15	288
AAR/RAM Transition - AAR/RAM Traj Sim.	0 0	97	32	465
RAM	28 (14*)	342 (246*)	21	325
RAM/SCRAM Transition	0	0	2	20
SCRAM	*	112*	58	1218
SCRAM/Rocket Transition	21*	279*	0	0
Rocket-Only	16	129	10	72

<sup>\*</sup> Direct-Connect Tests

#### **ART Future Plans**



## Fabricate flight weight components

Rocketdyne combustor

Aerojet combustor

Aerojet ceramic ram/scram injectors

Test selected components

Document ART project

ART is scheduled to conclude in 2001

#### RAPTOR

(Revolutionary Airbreathing Propulsion Technology demonstratOR)



## **Combined Cycle Propulsion Testbed**



- Take the next logical step in combined cycle propulsion development
- Develop a flight-weight rocket based combined cycle engine system ground testbed
- Sized capable of accelerating a self powered vehicle from Mach 0.8 to Mach 7
- for air-augmented rocket, ramjet, and scramjet modes Demonstrate RBCC engine system operation
- Provide testbed for evaluation of candidate innovative components
- Demonstrate flight weight engine system design and fabrication
- Evaluate engine system operational characteristics
- In test in 2004
- A testbed, not a prototype of an operational engine
- operational engine development(s) beginning in the 2010 2015 One of several airbreathing engine system testbeds leading to

## **Hydrocarbon Demonstrator Tracebility**



### Provides Tracebility to an Operational Launch Vehicle by .. The Hydrocarbon RBCC Engine Systems Demonstrator

- Developing a flight like, thermal & power balanced RBCC engine system
- Demonstrating the operation of an RBCC engine system by testing from Mach 0 through Mach 7 in ground test
- Performing vehicle design and propulsion system integration studies to show the applicability of RBCC to earth-to-orbit propulsion systems

#### **On-Going Activites**



## Industry Team is Key to Development - HYPAR

- Preserve U.S. high speed propulsion industrial base
- Rocketdyne Management Lead
- Pratt & Whitney Technical Lead
- Aerojet Systems Integration Lead
- MOA signed
- FTC concurrence 8/4
- Teaming agreement to be signed by 10/15
- Program planning underway
- Engine System Study final report week of 10/30



#### Flowpath Selection Team

- Team has been convening since June
- Two representatives from each of the engine companies

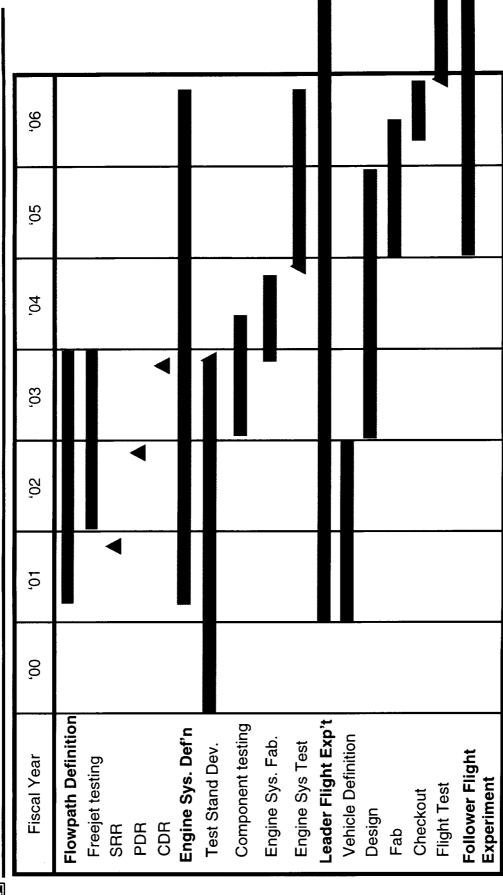
AEROJET

- One representative from Boeing Phantom Works
- One representative from each participating NASA center (DFRC, GRC, LaRC, MSFC)
- Data sharing initiated 7/24
- Selection made 9/1 Aerojet Strutjet Flowpath



- ◆ Perform planning activity 11/00 4/01
- Put HYPAR consortium under contract
- Demonstrator vehicle activity led by LaRC
- Feed requirements for engine system

## **Long Term RAPTOR Schedule**



## Flight experiments not part of the RAPTOR project



#### Milestone/Activities



#### Milestones

- 10/00 Engine system study final report
- 9/01 System Requirements Review (SRR)

### Prioritized List of Activities

- Perform detailed project planning
- Implement system engineering framework
- Refine selected flowpath, engine systems concept, and structural approach
- Itterate propulsion/airframe integration